





Development of ADRIAN Joint Analysis Software Methods and implementation

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M. Sc. Thesis

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Gothenburg, Sweden 2002

Preface

This thesis concludes the authors' M.Sc. degree in Civil Engineering at Chalmers University of Technology, Gothenburg. The thesis work has been carried out at Volvo Car Corporation. Nicklas Bylund at the department of Painted body Engineering has supervised the work. Examiner has been Klas Samuelsson from Finite Element Centre at Chalmers University of Technology.

We would like to thank our supervisor and our examiner for all help and support during this thesis work. We would also like to thank all people involved at Volvo Car Corporation, especially Jan Samuelsson for all help regarding the general analysis program MSC/Nastran.

Gothenburg, March 2002

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Abstract

The background to this thesis work is the lack of software analysing car body joints. In order to shorten car development lead-time, it is important to control properties of car body elements. Studying the present development process, design engineers have little insight in the mechanical properties of their produced body parts. As engineering analysts enter the scene, the development process has come to a state where radical changes are hard and costly to achieve. Thus, there is an information gap between the concept model made by senior engineers experienced with analysis tools and the detail design made by design engineers.

As a short development lead-time is of great interest in the automotive industry of today, a main goal is to involve the design engineers in the area of engineering analysis. As creating models for finite element analysis is quite complex, the design engineers do not have the time to learn and use available analysis software. The solution applied here is development of a single purpose program, which should be time saving and easy-to-use.

This problem formulation resulted in new software for joint analysis called ADRIAN. ADRIAN imports a geometry file in the IGES-format, transforms it into a finite element model, submits a job to the solver MSC/Nastran and visualises the results by means of a frequency animation and an HTML-page giving information on stiffness values for the joint.

The design of the software is focused on the intended users. Therefore most of the interactive job is located to the home environment of the design engineers, i.e. the CAD-software Catia. After having exported the joint model from Catia, the remaining calculation process should only be a few interactions away. The presented process and tool is not intended to replace the complete body simulations of today, but merely permitting the design engineer to make a preliminary check of his design, leaving the detailed analysis to the analysis department.

The conclusion is that ADRIAN can be very useful for controlling car body joint design and in giving design engineers a deeper understanding for joint stiffness aspects. In turn, this may free resources in engineering analysis departments or make expensive consultant cases regarding joint stability unnecessary. By creating all joint models with the same algorithm, comparing results becomes more interesting.

A comparison of models created by ADRIAN and a job performed by the technical consultant company Semcon was made. Due to differences in modelling methods, comparing absolute frequency values would be misleading. Comparing the shape of the first three modes, the results generated by ADRIAN were identical to Semcon results.

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1 Introduction

This report is divided into two parts. The first part (chapter 3-6) contains a description of methods used today for evaluating joint capacity. This part will also describe the method developed in this thesis work, and implemented in the new software called ADRIAN. The second part (chapter 7) describes how the software works. The software is constructed using C++.

1.1 Background

In the present development process, design engineers have little insight in the mechanical properties of their produced joints. The present situation with analysts following up the work of the design engineers is somewhat ineffective. It may also be difficult to achieve radical changes for the analysts, as the work done by the design engineers is followed up late in the development process, making changes costly.

Today different crash situations are simulated using finite element models of complete cars. In 2001, attempts to strengthen the bond between performance of the complete car and performance of single components were made. This resulted in the software DAMIDA, used for section capacity simulation. This program in addition with software focused on joint analysis would form an effective tool in controlling car body architecture.

1.2 Purpose

The purpose of this thesis work is to bridge the gap between the concept model made by senior engineers experienced with analysis tools and the detail design made by design engineers. By revealing joint design problems early in the development process, the risk for late radical design changes is reduced.

<u>1.3 Goal</u>

The goal of the project is to introduce software capable of analysing a joint with few user interactions. The design engineers are the intended users of the software. By using the software, the design engineers shall have the possibility of making a preliminary check of the joint design.

2 Problem formulation

In order to reach the goal described in the previous section, the software must be designed for the specific platform and users at Volvo. The main features of the software are capability to import a joint Catia-model via the graphical format IGES, transformation to a finite-element-model, linear analysis and visualisation of stiffness and frequency aspects of the joint.

2.1 Program specification

The resulting program is intended to have characteristics as follows:

- Simple joint model construction in Catia
- Compatibility with the graphical exchange format IGES
- Easy-to-use graphical interface
- Automatic geometry meshing
- Automatic and mesh-independent adding of weld spots
- Automatic modelling of joint leg ends
- Automatic submission to the MSC/Nastran solver
- Automatic and pedagogical presentation of results

3 The role of the Finite Element Method

3.1 Engineering analysis methods

Engineering analysis can be divided into two different categories, classical methods and numerical methods.



Figure 3.1: Engineering analysis methods according to [9].

3.2 Classical methods

Exact classical methods can only be applied on simple geometries, though they attempt to solve differential equations based on fundamental physical properties. To increase the application range, problems can be solved using approximate solutions to the differential equations. This approach has its advantage in a high degree of problem insight, but is still way to narrow to be applied on the geometric shapes typical for car body construction [9].

3.3 Numerical methods

There are different numerical methods, divided by the properties the method focuses on. Common for all the methods is that they address a broad range of problems. The energy method minimises the potential energy for a structure over its entire domain. This is very well suited for some problems, but it is not broadly applicable. The boundary element method approximates functions satisfying the governing differential equations, except for the boundary conditions. By letting elements represent only the boundary of the domain, problem size is reduced. This method has its limitation in the fact that the solution to the governing equations can be hard to obtain. The finite difference method replaces governing differential equations and boundary conditions with corresponding algebraic equations. This makes solution of somewhat irregular problems possible, but complex geometry, boundary conditions and load become difficult to handle [9].

3.4 Finite Element Method (FEM)

The Finite Element Method is the most general numerical method, and often the only possible method of analysis. The problem generality of the method is virtually unlimited. This is achieved by using basic elements of regular shapes to represent the geometry. These elements can be combined to approximate any irregular boundary. The method seeks to approximate the behaviour of an arbitrarily shaped structure under general loading and constraint conditions with an assembly of discrete finite elements. By taking the collective behaviour of the elements into account, the behaviour of the entire structure is obtained [9].

3.5 Linear analysis

When modelling the entire structure as an assembly of elements, a stiffness matrix represents each element. The total stiffness matrix derived from all contributions from the element stiffness matrices is the so-called global stiffness matrix K. Letting k represent the stiffness matrix for each element, F is the vector of forces and u is the vector of displacements resulting from F. The equation of movement is,

$$F_{elem} = k_{elem} u_{elem} + m_{elem} u_{elem}$$
(1)

As all contributing elements describe the behaviour of the entire structure, the equation for the structure is,

$$F = Ku + M \ddot{u}$$
 (2)

This is the equation solved by MSC/Nastran in linear analysis. With zero applied force,

 $Ku + M\ddot{u} = 0 \tag{3}$

Looking for solutions of the form,

 $u = u_0 \cos(\omega t) \tag{4}$

We obtain the vibration equation,

 $Ku = \lambda Mu$ (5) where $\lambda = \omega^2$ Thus, solving equation (5) becomes a matter of inverting the matrix K with an effective algorithm. The solution flowchart for the method is shown in figure 3.2.



Fig 3.2: Solution flowchart for Finite Element Method [9]

Assumptions and limitations of linear analysis

There are a few assumptions and limitations in linear analysis which must be taken into account when using this solution method. Failure to ensure that these restrictions are accounted for will lead to solutions not representing the factual physical properties of the structure [9].

- Linear elastic material: Material must be isotropic and homogeneous. The method assumes material in which stress is directly proportional to strain. Loads may not take the material beyond the point where the material is no longer elastic. The unloaded structure must also be free of initial and residual stress.
- **Small displacements:** The method is restricted to small displacements. This restriction is due to the formulation of governing equations for the elements. A lateral plate deflection can be considered small if it is substantially smaller than the thickness of the plate and a beam deflection substantially less that the smallest dimension of the cross section of the beam.
- **Slowly applied loads:** The structure must be in static equilibrium. Loads may not induce a dynamic effect. Therefore loads disturbing this equilibrium, for example impact loads, are not allowed. Slowly applied loads shall be taken to mean loads that do not result in a significant dynamic behaviour.

4 Methods for analysing joint capacity

4.1 Experimental methods

Background

In the seventies, Lubkin showed that the overall stiffness of frame structures are very sensible to the stiffness of joints, governing up to 60 % of the global stiffness. Later, Moon evaluated 3-legged joints in the plane by representing the joints analytically with rigid elements and torsion springs. The joint is then measured statically in order to identify the stiffness of the springs [3].

Problems

When evaluating joints experimentally there are two major difficulties. Problems occur when trying to define the centre of the joint and in the area of experimental set up. Due to complexity in car body design, joints may have more than one centre, as in figure 4.1.



Figure 4.1: Joint without an absolute centre [3]



Figure 4.2: Measurement set up [3]

Parametrising a joint with a lack of a well-defined centre using rigid elements and torsion springs is unfeasible. The legs can not be represented unambiguously leading to great errors. Even when a joint centre can be defined, there are problems with the experimental set up. The joint has to be permuted, i.e. each leg has to be grounded while one of the others are attached to a force lever [3], see figure 4.2.

An alternative approach

In order to avoid these problems, an alternative set up method can be applied. The joint is set up freely using soft springs and rubber ropes and heavy masses are cast to the joint legs. The joint is defined as inscribed in a sphere with radius r, for an automotive joint 250 mm. The centre of the sphere is put at the mean distance from the different joint centres. The joint is cut out where the sphere cuts the legs (figure 4.3).



Figure 4.3: Alt. measurement method

Dynamic measurements are carried out with hammer impact and accelerometers on various locations on the joint. The frequencies and the corresponding eigenvectors are used to visualise the vibrational behaviour of the joint [3].

This method has been introduced at Volvo. Thus, it is important to make virtual modelling similar to the models used experimentally in order to make results comparable.

4.2 Present simulation method

At this moment, there is no software available for simulation analysis of car body joints. The analysis is carried out manually either by engineers employed at Volvo or by engaging consultants. The manual method used by engineering analysts at Volvo is similar to the method used by ADRIAN.

4.3 Method used by ADRIAN

The software ADRIAN can be regarded as a "container of knowledge", where program components are contributions from different Volvo Car Company departments.

The Process

The design engineer exports a CAD-model and converts it into the IGESformat. The CAD-model contains information on joint geometry, weld spots, sheet thickness and weld diameter. There are also planes defining the end of the joint in the CAD-model. Information is separated in the Catia model by placing it in different sets. The material is not explicitly involved in the calculations. As ADRIAN uses linear analysis the material only enters in form of the ratio of the E-module and the density. This ratio is close to constant for the materials of interest.

The program consists of a pre-processor converting the IGES-model to the Nastran-format. Both the IGES-format and the Nastran-format are ASCII (plain text) formats. ADRIAN extracts the sets in the IGES-file and takes appropriate actions due to the set contents (see section 7.4).

The program ADRIAN has a rather simple user interface. ADRIAN is intended to bridge the gap between existing software used at Volvo. By using existent software in an efficient way, there is no need for an advanced interface with a great deal of user interactions.

As pre-processing is finished, ADRIAN submits a job to the solver MSC/Nastran. The solving process is carried out on an external server. Calculations are not treated on the local computer due to the fact that the local system configuration may differ between computers at Volvo. While processing, ADRIAN waits for significant files to be returned and then launches the post-processing sequence automatically. As post-processing is finished, the user obtains a HTML-page containing rotational and translational stiffnesses for the joint legs. ADRIAN also gives the possibility of analysing the frequency modes in the post-processing software Animator. These results can be used for a comparison with a full simulation of concept models.



Figure 4.4: The ADRIAN Method

5 Data formats

5.1 The IGES format

IGES (The Initial Graphics Exchange Specification) is a widely adopted standard for geometric data produced by CAD/CAE systems. IGES consist of five sections, the start section, the global section, the directory section, the parameter section and the termination section [17].

- **The Start Section:** This section contains information assisting the receiving user such as features and the revision number of the originating system.
- **The Global Section:** Here the parameters necessary for translating the file are given.
- **The Directory Section:** Contains an entry for each entity in the file comprising a code representing the entity type and sub-type and pointers to the entity data in the next section. Each entry consists of two lines comprising 18 fields with 8 characters used in every field.
- **The Parameter Section:** Contains the entity specific data such as coordinate values, annotation text, number of spline data points and so on. The first parameter in each entry identifies the entity type. From this parameter the meaning of the remaining parameters can be derived.
- **The Termination Section:** Marks the end of the data file, and contains sub-totals of records for data transmission check purposes.

By using IGES, ADRIAN is not restricted to be used in a Catia environment. The IGES entities identified by the program are given in table 5.1.

Туре	Entity	Туре	Entity	
102	Composite curve	128	NURBS surface	
108	Plane	142	Curve on parametric surface	
116	Point	144	Trimmed surface	
126	NURBS Curve	402	Set	
T 1 1 5 1 10 50 ("" 1				

|--|

More information on the algorithm used when reading IGES can be found in section 7.4. An example of an IGES-file can be found in Appendix B.

5.2 The Nastran format

The Nastran format is the text format representing FEM models that are to be submitted to the MSC/Nastran solver. Data is given in blocks consisting of eight characters or by separating blocks by a comma. ADRIAN represents data in blocks of eight characters. The first data block specifies the element type, for example 'GRID' is a grid point in the mesh and 'CQUAD4' represents a mesh element with four edges. The element type specifies the syntax of the parameters belonging to the element. An example of a Nastran-file can be found in Appendix C.

6 Description of Software used by ADRIAN

<u>6.1 Catia</u>

Catia is the CAD software used by Volvo. CAD (Computer Aided Design) is used for creating a digital product definition. The car body components are drawn digitally in Catia. Volvo uses shell modelling, i.e. sheets are represented by a surface which is given a theoretical thickness. Weld spots are in separate models, consisting of weld spots for the entire car.

6.2 ANSA

ANSA (Automatic Net-generation for Structural Analysis) is the pre-processing and meshing software used by Volvo. The program contains geometry cleanup-, meshing- and mesh improvement-functions used by ADRIAN. ANSA has the possibility of submitting script jobs to the program. ADRIAN creates scripts that are read by ANSA on each run. ANSA consists of an IGES translator and a Nastran-export routine. ANSA also supports STEP (Standard for the Exchange of Product model data). ADRIAN is implemented in a way making a future add of STEP-compatibility possible.

Mesh and mesh improvement

ADRIAN uses the ANSA Free-meshing-algorithm. This algorithm creates the least number possible of elements, taking into account the shell element distortion and minimum length.

Element quality checks in ANSA are made by means of aspect ratio, warping, skewness, taper and jacobian (figure 6.1).



Figure 6.1: Element quality criteria in ANSA [2]

In the ANSA version 11.1, the mesh improvement algorithms supported by the ANSA scripting language used by ADRIAN are algorithms for reducing element warping and jacobian [1].

6.3 MSC/Nastran

MSC/Nastran is the Finite Element problem solver. This section describes the method used by Nastran for solving the linear FEM-problem [11].

The Lanczos Method

Nastran uses a solution method called the Lanczos method. The Basic Lanczos recurrence (1950) is a transformation process to tridiagonal matrix form.

Basic Lanczos Recurrence

The basic Lanczos recurrence solves the eigenvalue problem, $Ax = \lambda x$ (2.1) and can be formulated as:

1. Initiation

- a. A starting vector q_1 is chosen with $||q_1|| = 1$.
- b. $\beta_1 = 0$ and $q_0 = 0$ are set.
- 2. Iteration

For j=1,2,3...., iterate until convergence as follows:

$$r_{j+1} = Aq_j - \alpha_j q_j - \beta_j q_{j-1}$$

where,
$$\alpha_j = q_j^T Aq_j$$

$$\beta_{j+1} = \left\| r_{j+1} \right\|$$

$$q_{j+1} = \frac{r_{j+1}}{\beta_{j+1}}$$

If the algorithm is carried out without round-off errors, the vectors q_1 , q_2 ,... are orthonormal. These vectors can be considered to be column vectors of an orthogonal matrix $Q_j = (q_1 \quad q_2 \quad ... \quad q_j)$. The scalars α_j and β_j can be combined in a tridiagonal matrix T_j :

$$T_{j} = \begin{pmatrix} \alpha_{1} & \beta_{2} & & & \\ \beta_{2} & \alpha_{2} & \beta_{3} & & \\ & \beta_{3} & \dots & & \\ & & \beta_{i} & \alpha_{i} & \beta_{i} & \\ & & & & \dots & \beta_{j} \\ & & & & & \beta_{j} & \alpha_{j} \end{pmatrix}$$

The eigenvalues and corresponding eigenvectors of *A* are computed from those of T. Let θ and *s* be an eigenpair of T_i , thus

 $T_i s = s \theta$

then θ and y with $y = Q_j s$ is an approximate eigenpair for the original problem. This is where the computational gain can be found. In a typical application the order of the matrix A may be in the tens of thousands, whereas the order of T_i is about 20 to 30.

Modification of the Basic Lanczos Recurrence

In order to apply the Basic Lanczos Recurrence on the vibration problem, some modifications have to be made. The vibration problem is a symmetric generalised eigenvalue problem using the following form:

 $Kx = \lambda Mx$

where *K* is a symmetric matrix, *M* is a symmetric positive semidefinite matrix and λ is the eigenvalue.

If applying the Basic Lanczos Recurrence on the vibration problem, the algorithm yields precise approximations to the large, well-separated, but uninteresting, eigenvalues and poor approximations to the small, clustered, and interesting eigenvalues. By shifting and inverting the eigenvalue problem, the Lanczos algorithm can be applied with a better outcome:

$$M(K - \sigma M)^{-1}Mx = \frac{1}{\lambda - \sigma}Mx$$

The shift σ is chosen close to the eigenvalues of interest. A properly chosen shift guarantees rapid convergence to all eigenpairs in the vicinity of the shift. The cost is the factorisation of $(K - \sigma M)$.

The shifted Lanczos algorithm has difficulties with eigenvalues of high multiplicity. Each step of the shifted Lanczos recurrence requires the solution of a sparse linear system of equations of the form $(K - \sigma M)x$ and one multiplication by the matrix M. These operations require accessing matrices stored on disk file and thus entail significant I/O costs. Due to this fact, MSC/Nastran uses a Block Lanczos Algorithm. This approach makes multiple eigenvalues easier to compute and the amount of I/O is reduced. The result is the general Lanczos procedure.

The procedure used by MSC/Nastran

The general Lanczos procedure used by Nastran is summarised in figure 6.2 on the next page. The procedure consists of two levels, the outer level in shift strategy and administration, and the inner level is the actual Lanczos iteration.



Outer Level Lanczos Procedure

Inner Level Lanczos Procedure

Figure 6.2 The Lanczos Procedure [11]

Superelement reduction

In order to reduce the stiffness output matrix, the joint is modelled with a master node at each leg. Dividing the stiffness matrix K in a masternode part and in a part describing internal interaction, only substantial information on the stiffness characteristics of each leg is obtained. If K_{mm} is the stiffness from interaction between masternodes and K_{ss} is the stiffness from interaction between internal nodes we have that

 $\begin{pmatrix} K_{mm} & K_{ms} \\ K_{ms}^T & K_{ss} \end{pmatrix} \cdot \begin{pmatrix} D_m \\ D_s \end{pmatrix} = \begin{pmatrix} F_m \\ F_s \end{pmatrix}$

where D_m are the degrees of freedom for the masternodes, D_s are the internal degrees of freedom, F_m are the forces on the masternodes and F_s are the forces on the internal nodes.

By taking

$$K_m = K_{mm} - K_{ms} K_{ss}^{-1} K_{ms}^T$$

the reduced stiffness matrix is obtained. In this way, only stiffness information describing the masternodes is given. Note that this reduction does not include any approximation of the original problem. [10].

The stiffness output

The result of the reduction described above is a 18x18 stiffness matrix for a joint with three legs, and a 24x24 matrix for a joint with four legs, represented in the structure shown below:

$$K = \begin{pmatrix} \sigma_{xx}^{T1} & \sigma_{xy}^{T1} & \sigma_{xz}^{T1} & \dots & & \\ \sigma_{yx}^{T1} & \sigma_{yy}^{T1} & \sigma_{yz}^{T1} & \dots & & \\ \sigma_{zx}^{T1} & \sigma_{zy}^{T1} & \sigma_{zz}^{T1} & \dots & \dots & \dots & \\ \dots & \dots & \sigma_{xx}^{R1} & \sigma_{xy}^{R1} & \sigma_{xz}^{R1} & \dots & \\ & \dots & \sigma_{yx}^{R1} & \sigma_{yy}^{R1} & \sigma_{zz}^{R1} & \dots & \\ & \dots & \sigma_{zx}^{R1} & \sigma_{zy}^{R1} & \sigma_{zz}^{R1} & \dots & \\ & & \dots & \sigma_{zx}^{T2} & \sigma_{xy}^{T2} & \dots & \\ & & & \sigma_{zy}^{T2} & \sigma_{zz}^{Rn} & \\ & & & & & \sigma_{zy}^{Rn} & \sigma_{zz}^{Rn} \end{pmatrix}$$

where for example $\sigma_{yy}^{T_2}$ represent the translational (T) stiffness for masternode 2 in the y-axis direction.

The elements outside the tridiagonal are stiffness values representing the connection between the translational and rotational stiffness and the connection between different masternodes. These elements are not of interest for ADRIAN post-process purposes. Each 3x3 submatrix represent a translational and rotational tensor for the specific masternode. This tensor describes the stiffness values in all directions for the node.

The frequency output

Animations of the 12 first modes are stored in a file, named .op2, by the Nastran solver. The modes are described by the eigenvector calculated by Nastran. The frequency of each mode can be found in a so called f06-file. Each frequency is the square root value of the corresponding eigenvalue,

$$f = \frac{\sqrt{\lambda}}{2\pi}$$

Thus, small eigenvalues correspond to low frequency. The low frequency modes are the most interesting ones, describing the weakest directions of the joint.

6.4 Animator

Animator is an graphic post-processor for finite element computations. It can be used for different finite element applications, such as crash and static. ADRIAN uses Animator in order to animate frequency modes.

7 Program description

This chapter contains a description of the developed program.

7.1 Program requirements

ADRIAN is developed using the HP-UX 11.0 platform. The program uses existent software available at Volvo. ADRIAN is tested with the configuration described below:

- HP-UX 11.0
- Catia v. 4
- ANSA v. 11.1
- MSC/Nastran v. 2001
- Matlab v. 5.3
- Animator v. 5.3

7.2 File structure

Directory creation

The first time ADRIAN is started, the program creates a directory named ADRIAN_ROOT in the user home directory. Under this directory a directory named .TEMP is also created.

Each time the program is executed, a temporary working directory for this specific session is created. This temporary working directory is named according to the time ADRIAN was executed. By using separate temporary working directories, it is possible for the same user to start many independent ADRIAN sessions.



Figure 7.1: File structure used by ADRIAN

When the pre-processing sequence is running, ADRIAN stores temporary files in the temporary working directory. As pre-processing is finished, a result file directory is created, named accordingly to the model imported and a job identification number given by the program. The temporary working directory is also used for storing temporary files used by the post-processing routines. By storing all ADRIAN files under the same directory, result-files could easily be removed if desired.

Directory deletion

If the user chooses to quit the ADRIAN session, the temporary working directory is deleted. This is done regardless the user quits by entering the file/quit menu or presses CTRL-C. In order to make sure no unwanted old temporary directories are left behind, the program checks for temporary working directories older than one day when a new session is executed and deletes them.

7.3 Graphical User Interface

The ADRIAN Graphical User Interface (GUI) is programmed using OSF/Motif. As most software for GUI design requires installation of libraries, the GUI was programmed directly in Motif. Motif libraries are in the default installation at Volvo. The GUI and the program functions are well-separated, leaving the possibility of attaching the program functions to another GUI in the future.





Figure 7.2: The ADRIAN Graphic User Interface

Figure 7.3: Settings menu

The program may be run completely or by launching separate routines. The complete process consists of pre-processing, Nastran-solving and post-processing. Each of these sub-processes is stand-alone and may be executed from specific menus. Under the options menu, settings for the process are set. There is also a help section available from the interface, consisting of relevant user information. The ADRIAN Help can be found in Appendix E. In the Output-window, processing information is printed. Examples of given information are IGES-file contents, Nastran calculation results and processing errors. During processing, the ADRIAN GUI is locked from interaction.

7.4 Pre-processing

Joint modelling in Catia

This section describes what to export from Catia. To use ADRIAN successfully it is important to use the syntax described below. It is also very important that the model exported is clean and therefore only consists of the components described below:

- Each sheet shall be placed in a separate Catia set
- A separate set consisting of **planes** describing the beam ends must be existent
- Sets for the welds

Sheets

Each sheet must be placed in a separate set named according to the syntax: **NAME**:T**SHEETTHICKNESS**. A setname may consist of maximum 30 characters. **NAME** is a unique name for each set, for example SET1, SET2 and SET3. The second part, **SHEETTHICKNESS**, is the sheet thickness given in millimetres in the range 0.5-2.5 mm. The sheet thickness is optional, if no value is given, ADRIAN uses a default sheet thickness of 1.0 mm. The following setnames are examples of valid ADRIAN setnames: SET1:T2.1, SET2:T0.9 and SET3.

Restrictions on geometry

To obtain the best performance of ADRIAN, the model should have faces on every surface. This is discussed more thoroughly later in this section.

Planes

A separate set consisting of planes describing the joint ends must be existent. ADRIAN looks for a set consisting of three or four planes, depending on joint type. The setname must be unique and consist of a maximum of 30 characters. There must be nothing else than these planes in the plane set.

Welds

In order to weld two sets together, a new set named

NAME:NAME:DDIAMETER is created, where NAME is the name part of the sheet set and DIAMETER is the weld diameter given in millimetres and in the range 4.0-9.9 mm, for example: SET1:SET2:D4.5. It is also possible to weld three or four sets together, for example: SET1:SET2:SET3:D6.3 or SET1:SET2:SET3:SET4:D5.5. The



Figure 7.4: A Catia joint model

weld diameter is optional; it is possible to create a weld set with the syntax **NAME:NAME**, for example SET1:SET2. If this is done, ADRIAN uses a default weld diameter of 4.5 mm. As the set is created, points marking the position of the welds are placed in the set.

If, for some reason, ADRIAN fails to create a weld, the user gets a warning message identifying the point causing the problem. It is important to make sure weld spots are in logic positions and that they belong to the correct sets, named after the sheets being welded together. There must be nothing else than points in a weld set and there is no limit in how many weld sets that can be existent in a model.

Reading IGES

ADRIAN consists of an algorithm for reading the IGES format. IGES indata is restricted to have the characteristics as described above. The program checks that these restrictions are fulfilled and takes actions with respect to the content identified.

Reading sets

As mentioned above, the information the program needs shall be stored in separate sets. The program identifies these sets in the IGES-file and appends them to a linked list. When this is done, the program checks the contents of each set. In order to continue processing, ADRIAN must identify a set containing planes describing the joint leg ends, at least one set containing geometry (i.e. at least one sheet) and at least one set containing weld spots.

- Planes: ADRIAN looks for a set containing three or four planes and nothing else. Thus, the software is capable of analysing joints with three or four legs. As the set of planes is identified, ADRIAN stores these planes to a temporary file.
- **Geometry:** The must be at least one set consisting of IGES NURBS surfaces or IGES Trimmed Parametric surfaces in the IGES-file. One set containing geometry is equal to a single sheet. As a set containing geometry is found, ADRIAN creates a separate IGES-file with the set contents.
- Welds: Weld spots are represented in sets containing points only. The set name shall consist of information on the sheets that are to be welded together and the weld diameter. The welding information is stored to a file used by the welding program SPOT.

Reading geometry sets

For each entity in the geometry set, ADRIAN checks if the entity is a Trimmed Parametric Surface or a NURBS Surface. An example of a line representing a Trimmed Parametric Surface in the Parameter Section can be found on the next page.

102,4,201,203,205,207,0,0;	209P	375
142,0,165,199,209,0,0,0;	211P	376
144,165,1,1,189,211,0,1,215;	213P	377
Figure 7.5: Trimmed Parametric Surface representation		

The first parameter, the integer 144, indicates a Trimmed Parametric Surface. Then there is a pointer to the definition line of the surface entity D that is to be trimmed (165). The third parameter indicates whether the outer boundary of the D surface is the boundary of the trimmed surface. In this case the parameter is equal to one indicating that the outer boundary of D is not the outer boundary of the trimmed surface. The next parameter is an integer telling the number of inner boundaries. Then there is a parameter pointing to the Curve on a Parametric Surface Entity that constitutes the outer boundary (189). The succeeding parameters (in this case only the integer 211) are pointers to the inner boundary curves.

If there are faces on all surfaces in the Catia model the only geometric representation in the IGES-file will be Trimmed Parametric Surfaces. Catia surfaces which have not a corresponding face are represented by IGES NURBS surfaces (Entity ID 128). If ADRIAN finds single surfaces in the model, the user is confronted with a warning message.



Figure 7.6: Warning message for models consisting of surfaces which do not have a corresponding face.

It is strongly recommended that models containing faces on all surfaces are constructed. If this warning message occurs when importing such a model, the user knows that the surfaces detected are not intended to be included in the model. The user simply clicks the Remove-button to clean the model from these unwanted surfaces. Moreover, performance of translating IGES surfaces in ANSA is better for a model consisting of faces on all surfaces (Trimmed Parametric Surfaces) than for a model containing single surfaces (NURBS Surfaces).

Using ANSA

As the information needed is identified, ADRIAN starts processing the data. ANSA has the capability of launching script jobs. As all sets containing geometry have been found, ADRIAN creates a script which is sent to ANSA. The script contains information on what IGES-files to be read and what to do with each file. ADRIAN splits the total geometry in files consisting of only one sheet due to the fact that problems might occur in ANSA if two different sheets are close to each other. In this case, surfaces belonging to different sheets may be connected to each other.



Figure 7.7: Surface boundary representation in ANSA

In figure 7.7 yellow lines represent a connected surface boundary and red line represent an unconnected outer boundary. If a red line can be detected inside the outer boundary of the sheet, ANSA has failed in translating CAD-data correctly and two neighbouring surfaces are unconnected. If a blue line is existent, more than two surface boundaries are inside the tolerance for connection. For the program purposes there shall not be any blue lines existent when reading data into ANSA.

The meshing process

Each sheet is translated and a mesh is constructed. As ADRIAN is intended to be used with a low amount of user interactivity, the default mesh size is set to 4 mm. This is regarded as a very small mesh size and increases the time for solution of the FEM-problem. However, the advantage concerning a reduced risk of badly shaped elements is a strong argument for a fine mesh. Increasing the mesh size increases the risk of getting elements not tolerated by the Nastran solver, for example triangulary shaped quad-elements.

As meshing is finished, ANSA makes some slight improvements of the mesh. The software consists of algorithms for detecting bad elements and improving the representation, checking quality factors as described in sec 6.2. The improvement algorithms used are the Reduce/Split and the Reduce/Jacobian algorithms. By using the Split algorithm, quadrangular elements with high amount of warping are splitted, creating two triangular elements. Triangular elements can not be warped though a triangular element is always planar. The Jacobian algorithm reduces the number of second order elements exceeding the Jacobian quality criteria.

No licence

The number of ANSA licences is limited. If ADRIAN is unable to start ANSA due to a lack of licences, pre-processing is interrupted and the user is recommended to make a new attempt later on.

Welding

A requirement for modelling welds automatically and in an efficient manner is that the welds are constructed independent of the mesh. To accomplish this, a program designed for modelling weld spots is used. By transforming the information from the IGES-file, i.e. weld spot coordinates, weld diameter and what sheets that are to be connected, suitable indata for the welding program SPOT is printed to file by ADRIAN. The sheets are identified by their Nastran PID (Part ID).

SPOT creates a solid hexa element (CHEXA) representing the spot weld itself, and connects the corners of the solid element to neighbouring mesh elements. SPOT uses the displacement interpolation functions of the CQUAD4 and CTRIA3 elements (Mesh elements with three or four corners). For each element, each displacement component (including rotation degrees of freedom) is represented by a bi-linear shape function. The problem is to determine the contribution of each corner point of the CQUAD4 or CTRIA3 to the displacement of a point on the interior of the element. SPOT solves the problem by manipulating the shape functions. At the end of this process the following equations are obtained for one of the solid element corners with id 10002.

$$u_x^{10002} = a_1 u_x^8 + a_2 u_x^9 + a_3 u_x^{11} + a_4 u_x^{12}$$

$$u_y^{10002} = a_1 u_y^8 + a_2 u_y^9 + a_3 u_y^{11} + a_4 u_y^{12}$$

$$u_z^{10002} = a_1 u_z^8 + a_2 u_z^9 + a_3 u_z^{11} + a_4 u_z^{12}$$

where 8,9,11 and 12 are the corners of the neighbouring CQUAD4 element. Note that the coefficients are the same for the x,y and z displacements. This information is then expressed using a rigid element connecting the solid element corner with the mesh. These calculations are made for all eight corners of the solid element. The resulting geometry is shown in figure 7.9.





Figure 7.8: Weld spot structure

Figure 7.9: Weld spot between two flanges

The program supports welding 2-4 sheets together with one weld. In the case of a three joined flange, two hexa elements are generated, one between each of the flanges [5].

Leg end modelling

As ADRIAN uses superelementreduction, the end-sections of the joint leg ends must be modelled. The program uses the indata planes to identify the elements nodes lying in the end plane. Taking the mean value of the coordinates of these nodes, defines the position of a masternode for the end section.



Figure 7.10. Joint leg end section with a rigid body

As this is done, the calculated masternode is connected with the nodes lying in the section with a rigid body.

In order to construct a model similar to the joints used in the experimental methods described in section 3.1, masses must be attached to the leg ends. By using a point mass with a given moment of inertia, a disc-shaped mass can be modelled. In ADRIAN this is done primarily by defining a local coordinate system for the joint leg. A Cartesian coordinate system is placed with its origo in the masternode and with the z-axis pointing in the direction of the normal vector of the plane. A mass is defined with its centre of gravity in the masternode and with moments of inertia in the x-,y- and z-axis directions. The mass is set to 5 kilos and the moment of inertia is calculated according to fundamental mechanical formulas for a solid cylinder:



Figure 7.11: Moment of inertia of a solid cylinder [13]

The mass is assumed to have r=125 mm. The density ρ is set to the density of steel i.e $\rho = 7.85 \cdot 10^{-9} \frac{tons}{mm^3}$. This result in $I_x = 19.7 tons \cdot mm^2$ and $I_z = 39.0 tons \cdot mm^2$.

Pre-processing options

The available options for pre-processing in ADRIAN is

- **Mesh size:** The mesh size used by ANSA can be set to a value of minimum 1 mm and maximum 20 mm. Default is 4 mm.
- Use Catia sheet thickness: • Sheet thickness is normally intended to be set in Catia. If the box is checked, ADRIAN searches for definitions of sheet thicknesses in the IGES-file. If no thickness is defined in the Catia model, ADRIAN sets the sheet thickness to the default value of 1.0 mm. If the box is unchecked, ADRIAN identifies the sheets in the IGES-file and creates a pop-up. In this pop-up the user may state the thicknesses.

- Set Thickness Enter value (0.5–2.5 mm)		
SET3	٥.80 mm	
SET2	[2.0 mm	
SET1	0.90 mm	
Ok		

Figure 7.12 Sheet thickness pop-up

- Use Catia weld diameter: Same as for the sheet thickness options but instead gives the possibility of changing actual weld spot diameters in the model.
- Advanced option/user interactivity in ANSA: This option is made in order to expand ADRIAN from only being a "black box" where indata goes in and results pop out. By unchecking the box, the program stops in ANSA for each sheet. The user can manipulate the geometry and then export the sheet in the Nastran format. This option is intended for people experienced with analysis and familiar with the software ANSA.



Figure 7.13: Advanced option

Storing information for post-processor

The pre-processor stores information for the ADRIAN stiffness post-processor in a post-file. This file contains joint geometry used for creating appropriate graphs.

Result from pre-processing



Figure 7.14: Resulting ADRIAN FEM-model

7.5 MSC/Nastran

MSC/Nastran jobs are submitted either automatically when running the complete process or by selecting the Nastran import-nas-menu. In order to perform a Nastran calculation from the ADRIAN menu, the model should have been pre-processed by the ADRIAN pre-processing routine.

As a job is submitted to the Nastran server, date and time is added to the post-file. This information is later given in the post-process sequence.

Nastran options

There are a few possibilities of controlling the tolerances of the solver:

- **Max Ratio:** The value of the Max Ratio indicates when Nastran shall consider the stiffness matrix to be singular. Default in ADRIAN is the same as the default for the solver (10⁷). As the diagonal ratio of the matrix increases this value calculation is interrupted.
- **Bail out:** If this box is checked, Nastran continues processing even if the stiffness matrix is found singular. This option can be useful when trying to identify the underlying problem to a high Max Ratio. For example, an unconnected part can be identified in the post-process animations.
- Queue: The default queue for the job in ADRIAN is csm_quick. This queue has a maximum of 10 minutes CPU time processing on the server. This should be enough for most joint configurations. The other queue available is the csm_small with a limit of 6 hours CPU time.

7.6 Post-processing

Visualisation of joint stiffness values

As Nastran has finished the job, an HTML-page will be constructed. This page contains information about the job and the most important graphs, such as a reference view of the joint and graphs representing the joint leg end-sections. The graphs of the end-sections contain information on the direction of the principal axes of stiffness and the absolute stiffness values in these directions. Information on the rotational stiffness of the joint leg can also be found here. The principles of the ADRIAN stiffness postprocessor are given in figure 7.15.



Figure 7.15: ADRIAN post-processing routines

The C++ post_punch-routine

The postpch-routine reads the ADRIAN post-file containing information on the model geometry and splits it into so called view.vw files. There are different view files for the masternode coordinates, the node coordinates for each sheet, the planes describing the joint ends, the nodes lying in each plane. This information is indata to the Matlab procedure. All data is ready for plotting. This is due to the fact that extracting data using loop algorithms is rather slow in Matlab.

The Unix script post_punch.sh

The C++ routine invokes an UNIX-script, creating an HTML-document and a MATLAB m-file. By running MATLAB in background-mode the result-files are evaluated and transformed into graphs. The MATLAB program extracts the stiffness matrix delivered by Nastran in the pch-format. In order to set a reference view for the graphs, three masternodes (there may be three or four) are chosen. These master nodes define a plane. The upward direction is set to the normal vector n_{up} to this plane. An orthogonal coordinate system is defined with the n_{up} vector along the z-axis. All nodes are transformed into

this coordinate system creating a reference view of the joint. As the reference view is created the following algorithm is repeated for each leg of the joint:

- Make sure that the normal of the plane (*n*_{end}) describing the end is pointing out from the geometry.
- Define the z-axis in a local orthogonal coordinate system by $z = n_{end}$.
- Define the x-axis in the local orthogonal coordinate system by evaluating $x = n_{up} \times n_{end}$.
- Define the y-axis in the coordinate system by $y = z \times x$.
- Transform all nodes lying in the plane describing the end to the local coordinate system and plot.
- Extract the translational and rotational stiffness tensor for the corresponding masternode from the total stiffness matrix. The tensor is of

the form
$$T_{end} = \begin{pmatrix} \sigma_{xx}^T & \sigma_{xy}^T & \sigma_{xz}^T \\ \sigma_{yx}^T & \sigma_{yy}^T & \sigma_{yz}^T \\ \sigma_{zx}^T & \sigma_{zy}^T & \sigma_{zz}^T \end{pmatrix}$$
.

• Define a normalised vector v in the plane and evaluate $\sigma^T = |v_s \cdot T_{end}|$ for all $v_s = v \cdot (\cos\theta + \sin\theta), 0 \le \theta \le 2 \cdot \pi$. Locate σ^T_{max} , σ^T_{min} and the corresponding direction v_s of the stiffness value. These two directions are

the so-called principal axes.Plot the principal axes.

• Evaluate the rotational stiffness around the z-axis by taking $\sigma^{R} = |z \cdot R_{end}|$.

The graphs and the resulting stiffness values are embedded in an HTMLdocument. This document can then be viewed in Netscape. By representing the stiffness values in the directions of the principal axes, the results can more easily be compared with, for example, optimisation results. Examples of HTML result-pages can be found in Appendix D.

Frequency mode animations

The second possibility of post-processing in ADRIAN is modal analysis. The post-processor is automatically launched when running a complete ADRIAN job but can also be invoked manually. The structure of the post-processor can be found in figure 7.15.

Animator

Animator is the most frequently used software for post-processing available at Volvo. ADRIAN uses Animator through the scripting language used by Animator. Many functions are predefined in Animator. In order to simplify the viewing of modes, ADRIAN defines the F1-F12 buttons for this purpose. The principle of definition of buttons is shown below:



Figure 7.16: Principle of definition of buttons in Animator

Before Animator is started, ADRIAN starts a UNIX-script named posta3.sh. This script creates an animation directory in the temporary working-directory. By invoking each animation from a unique directory, there is no limit in the

number of animations that can be run at a time. This script creates files f1.ses,f2.ses....f12.ses and mode1.ses, mode2.ses....mode10.ses in the created directory. The f.ses files refer to the corresponding mode.ses file. The mode.ses file contain information on which mode to be animated and the frequency of that mode. The frequency is taken from the f06-file generated by Nastran.

As Animator is started, the op2-file containing the animations is read in. Animator is started from the animation directory described above. Pressing F1 in Animator invokes the script f1.ses (Animator looks for this file in the directory the program was executed from), which in turn invokes the script



Figure 7.17: Post-processing in Animator

mode1.ses. The script mode1.ses starts the animation of mode 1 and prints the frequency of the mode to screen. The F12 button deletes the session files created and creates new sessions for the next ten modes. It also prints out what ten modes which are the currently available by pressing F1-F10. The F11 button is opposite to the F12 button, it toggles to the previous ten modes.

In Animator, the user can rotate and zoom the joint for a view in a different perspective. There is also a possibility of viewing nodal displacements in different colors.

No licence

If there is no Animator license when invoking the frequency animation postprocessor, ADRIAN gives a warning message and stops the post-processing sequence.

Post-processing options

There are two options for the ADRIAN post-process routines, both treating running ADRIAN in Complete process mode.

- **Do stiffness post-processing:** If this box is checked, ADRIAN launches the stiffness post-processor automatically when running in Complete mode.
- **Do frequency post-processing:** If this box is checked, ADRIAN launches the frequency animation post-processor automatically when running in Complete mode.

8 Program verification

8.1 Analysis of a symmetric joint

Joint design

In order to check the consistency of the calculation results, analysis of a

symmetric joint was carried out. A flanged sheet with a rectangular cross-section was drawn in CAD-software CATIA. The sheet was made bent, making an angle of 120°. This sheet was copied and rotated in space. By joining three parts of the design described above, a symmetric joint was constructed. On each flange eight weld spots was drawn, making a total of three weld sets and three sheet sets. The sheets were given a thickness of 1.0 mm



Figure 8.1: Symmetric joint

and the weld diameter was set to 4.5 mm. The total geometry is illustrated in figure 8.1.

Results



Figure 8.2: Results from ADRIAN analysis of a symmetric joint

Conclusions

The geometrical symmetry is reflected in symmetrical stiffness values. The principal axes for the three ends are identical, both by means of direction and absolute value. The rotational stiffness is same for all joint legs. This reduces the risk of a fatal error in the stiffness post-process algorithm.

8.2 Results from joint stiffness calculations of Volvo S80-model

Five joints from the Volvo S80-model were modelled in Catia with regards to ADRIAN standards. This in order to detect program problems and to compare the results with a consultant case carried out by Semcon in 2001-06-26. The global positions of the analysed joints are shown in figure 8.3.



Figure 8.3: Global positions of the analysed joints

Modelling method

The models were not made exact, i.e. they were constructed in order to verify the performance of ADRIAN. Sheet thicknesses were chosen according to drawing data. Weld spots were attached freely, no weld model was used. Sections were not cut according to the sphere-method described in section 3.1.

Results

The entire results, i.e. HTML-pages and animation pictures, can be found in Appendix D. These results are presented without absolute values. Results with absolute values included can be found in Appendix A.

Conclusions

The program showed good capability of transforming CAD-models to FEMmodels. Some faces were not connected in the models. The possible resulting problems are discussed below. The overall performance was very satisfying. For the most complex model, the time from submitting a job to a complete result was less than ten minutes. All of the FEM-models produced by ADRIAN passed through the standard limits for singularities set by MSC/Nastran.

The calculations of principal stiffness axes for the joint legs resulted in logical directions with the stiffest direction along the flange. Noteworthy is also that the joints positioned in the lower sections of the car body were stiffer than the upper joints.

Possible instabilities

When running ADRIAN, it is important that all the faces forming the entire sheet are successfully read when imported into ANSA. Sometimes ANSA fail to represent a face, remaining a hole in the mesh where the face should have been. Comparing results from a joint with a small hole and the joint in its correct representation, small holes do not seem to affect the global properties of the joint.

Instabilities can also be identified in the connections between the faces read in. ANSA has a tolerance level in order to define what faces are joined together. In one of the analysed joints, P23_B_LOWER, ANSA failed to connect two rather large faces. This resulted in a rip in the plate, visible when studying the modal animations. To fix the problem, cutting the face of interest in Catia may be one way. This may reduce the area containing the problem. Another approach is to use the advanced option in ADRIAN, correcting errors manually in ANSA. It may be valuable to seek the underlying problem causing the errors in ANSA, and, if possible, constructing a geometric CADrepresentation even more suitable for FEM-analysis.

However, it is important to emphasise that only one face of the thousands of faces in the models had a problem affecting the total result, and that this problem was easily corrected by cutting the face in Catia.

8.3 Comparisons with Semcon results

Due to differences in modelling method, comparing absolute values of frequency modes is not of great interest. For example, models differ in section cut position, mesh size, weld spot position and joint leg end mass modelation. A rough comparison can be carried out by comparing mode shapes [18].

Conclusion

The shape of the first three modes for each of the five models were equal to the mode shapes calculated by Semcon [18].
9 Future Work

Target values

In order to make analysing joints worthwhile, target values describing acceptable frequency and stiffness values for each joint in the car body is necessary. If ADRIAN is connected to the optimisation studies carried out on Volvo, appropriate target values can be achieved. As these values are set, it would be a good thing presenting the values in the HTML-result page, making comparison easy.

Modelling rules

Rules for construction of joint models in Catia must be set. If this is not done, comparison between models constructed by different design engineers will make no sense.

Data formats

It would be desirable if ADRIAN could import the Catia-model without an explicit transformation to IGES. This could be achieved in two ways. Either a more integrated IGES converter in future releases of Catia, or by adding the IGES-conversion script used by Volvo in the ADRIAN pre-processing sequence.

If the STEP-format become more frequently used in the development process at Volvo, it would be a good idea to implement ADRIAN STEP-compability.

Software performance

It is of great importance that some sort of quality check of the calculations made by ADRIAN is carried out in the future. This would increase the credibility in the method used by ADRIAN. This could be done by letting an experienced analysis engineer perform calculations on the joints presented in this report manually and by performing experimental tests using the method described in section 4.1. The results from this analysis can be compared with the results in Appendix A.

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10 Appendix

- A Flowcharts and source code
- B Example of IGES-file
- C Example of MSC/Nastran ecd-file and nas-file
- D HTML result documents and pictures of freqency animations for five joints in Volvo S80-series
- E ADRIAN Help section
- F Structural diagrams

Appendix A – Flowcharts and source code

ADRIAN flowcharts and source-code.

[Due to confidential restrictions from Volvo Car Corporation, the ADRIAN source-code and post-processing scripts are presented in a separate publication, Development of ADRIAN, Joint Analysis Software, APPENDIX A – Flowcharts and source code. This publication is only available at Volvo Car Corporation]

Appendix B - Example of IGES-file

An IGES-file consisting of a single surface with a corresponding face:

MODEL I ONEFACI	NAME : E	:					S S	1 2 3
1H,,1H	;,20F	HCATIA -	IGES PRO	DUCT,11H	HONEFACE.	IGS,57HIB	M CATIA IGES - CATIAG	1
SOLUT	IONS	V4RELEAS	E 2.2 RE	FRESH 01	,40HCATIA	A SOLUTIO	NS V4RELEASE 2.2 REFG	2
RESH 01	1,32,	75,6,75,	15,7HONE	FACE, 1.(),2,2HMM,1	1000,1.39	999999999999999999999,13H02G	3
0320.1	42321 N6	1	.0 <u>E</u> +04,/1	nrcr1086 N	0) CARS, 0, N	0 000020301D	4
4	06	0	Ő	1	1	0	PROP0001 0D	2
4	06	2	0	0	0	0	0 000020301D	3
4	06	0	0	1	15		PROP0015 0D	4
4	06	3	0	0	0	0	0 000020301D	5
4	06	0	0	5	1		PROP0001 0D	6
4	06	8	0	0	0	0	0 000020301D	0
4	06	9	0	0	10	0	0 00020301D	9
4	06	0	0	4	1	Ŭ	PROP0001 0D	10
4	06	13	0	0	0	0	0 000020301D	11
4	06	0	0	1	15		PROP0015 0D	12
4	06	14	0	0	0	0	0 000020301D	13
4	06	0	0	4	1	0	PROPUUU1 UD	14
4	06	18	0	1	15	0	DBOD0015 0D	15
4	06	19	0	0	10	0	0 00020301D	17
4	06	0	Ũ	2	1	Ŭ	PROPO001 0D	18
4	06	21	0	0	0	0	0 000020301D	19
4	06	0	0	1	15		PROP0015 OD	20
4	06	22	0	0	0	0	0 000020301D	21
4	06	0	0	2	1	0	PROP0001 0D	22
4	06	24	0	0	15	0	0 000020301D	23
4	06	25	0	0	10	0	0 00020301D	24
4	06	0	Ũ	3	1	Ŭ	PROPO001 0D	26
4	06	28	0	0	0	0	0 000020301D	27
4	06	0	0	1	15		PROP0015 0D	28
4	06	29	0	0	0	0	0 000020301D	29
4	06	0	0	3	1	0	PROP0001 0D	30
4	06 06	32	0	1	15	U	0 000020301D	31
4	06	33	0	0	10	0	0 00020301D	33
4	06	0	Ũ	2	1	Ŭ	PROPO001 0D	34
4	06	35	0	0	0	0	0 000020301D	35
4	06	0	0	1	15		PROP0015 OD	36
4	06	36	0	0	0	0	0 000020301D	37
4	06	0	0	3	1	0	PROPUUUI UD	38
4	06 06	39	0	1	15	0	PROP0015 0D	39 40
4	06	40	Õ	0	0	0	0 000020301D	41
4	06	0	0	2	1		PROP0001 0D	42
4	06	42	0	0	0	0	0 000020301D	43
4	06	0	0	1	15		PROP0015 OD	44
4	06	43	0	0	0	0	0 000020301D	45
4	06 06	45	0	2		0	0 000203010	40 47
4	06	0	0	1	15	0	PROP0015 0D	48
4	06	46	0	0	0	0	0 000020301D	49
4	06	0	0	5	1		PROP0001 0D	50
4	06	51	0	0	0	0	0 000020301D	51
4	06	0	0	1	15	0	PROP0015 OD	52
4	06	52	0	0	0	0	0 000020301D	53
4	06	58	0	0	⊥ ∩	0	0 000020301	54 55
4	06	0	0	1	15	5	PROP0015 00	56
4	06	59	0	0	0	0	0 000020301D	57
4	06	0	0	4	1		PROP0001 0D	58
4	06	63	0	0	0	0	0 000020301D	59
4	06	0	0	1	15	~	PROP0015 OD	60
4	06 06	64	0	U	U 1	U	U UUUU2U301D	61
4	00	68	0	4	⊥ ∩	0	0 000020301D	52 63
-1		00	0	0	0	0	5 000020001D	00

406	0	0	1	1.5		P	PROP0015 01	64
100	69	0	<u> </u>	10	0	0	0000203011	, 61) 65
400	0,0	0	0	1	0	U	0000203011	
406	70	0	4	1	0	F	(KOPUUUI UI	00
406	/3	0	0	0	0	0	0000203011) 6/
406	0	0	1	15		P	ROP0015 01) 68
406	74	0	0	0	0	0	0000203011) 69
406	0	0	2	1		P	PROP0001 01) 70
406	76	0	0	0	0	0	0000203011) 71
406	0	0	1	15		P	PROP0015 01) 72
406	77	0	0	0	0	0	0000203011) 73
106	0	Õ	12	1	0	Ū		, , , , , , , , , , , , , , , , , , , ,
400	0	0	12		0		0000001	, , , , , , , , , , , , , , , , , , , ,
406	89	0	0	0	0	0	0000203011	/ /5
406	0	0	T	15		P	PROPU015 01) 76
124	90	0	1	0	0	0	000000011) 77
124	250	8	1	0			AXS1 381) 78
124	91	0	0	0	0	0	0000102011) 79
124	0	0	1	0		Т	RANSFOR 01) 80
402	92	0	0	0	0	0	000003011) 81
102	0	0	1	16	0	Б		, 01 , 02
402	0	0	1	TO	0			02
406	93	0	0	0	0	0	0000203011	83
406	0	0	1	6004		P	PROP6004 01	84
124	94	0	0	0	0	0	0000102011) 85
124	0	0	1	0		Т	'RANSFOR 01) 86
406	95	0	0	0	0	0	0000203011	87
406	0	0	1	6004		P	PROP6004 01) 88
410	96	0	<u> </u>	0000	0	85	000020101	, 80 , 80
410	20	0	1	0	0	0.5	1 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	
410	0	0	1 Q	0	0	0	VUI 31	90
404	97	0	0	0	0	0	0000001011) 91
404	0	0	1	0			draft 41) 92
406	98	0	0	0	0	0	0000203011) 93
406	0	0	1	16		P	ROP0016 51	94
314	99	0	0	0	0	0	000002011) 95
314	0	0 0	1	0	0	0	COLOR12 01	96
214	100	0		0	0	0	00000201	, ,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
314	TOO	0	0	0	0	0	0000002011	97
314	0	0	1	0			COLORIA 01	98
314	101	0	0	0	0	0	0000002011) 99
314	0	0	1	0			COLOR15 0I) 100
128	102	0	2	61	0	0	0000300011) 101
128	250	-95	15	8			SUR29 91) 102
406	117	0	0	0	0	0	0000203011) 103
406	0	0	1	6007	÷	- P	PROP6007 01	104
100	110	0	1	104	0	0	000010501	101
126	118	0	1	124	0	0	0000105011	105
126	250	-99	ΤT	0			FAC50 111) 106
126	129	0	1	124	0	0	0000105011) 107
126	250	-99	7	0			FAC50 121) 108
126	136	0	1	124	0	0	0000105011) 109
126	250	-99	7	0			FAC50 131) 110
126	143	0	1	124	0	0	0000105011) 111
126	250	-99	- 7		0	0	EAC50 1/1	112
102	150	0	,	124	0	0	000010501) <u>112</u>
102	130	0	1	124	0	0	0000103011	113
102	250	-99	1	0			FAC5U 151) 114
126	151	0	1	124	0	0	0000100011) 115
126	250	-99	14	0			CRV133 161) 116
126	165	0	1	124	0	0	0000100011) 117
126	250	-99	10	0			CRV132 171) 118
126	175	0	1	124	0	0	000010001) 119
126	250	-99	10		0	0	CDV13/ 181	120
120	200	- 9 9	10	104	0	0	CRVI34 101	120
126	185	0	1	124	0	0	0000100011) 121
126	250	-99	9	0			CRV136 191) 122
102	194	0	1	124	0	0	0000100001) 123
102	250	-99	1	0			FAC50 161) 124
142	195	0	1	124	0	0	0000100011) 125
142	250	-99	1	0			FAC50 161	126
112	196	0	1	124	0	0	000020001	120
144	190	0	1	124	0	0	D000200011	120
144	250	-99	Ţ	U	~	-	PACOU 91) 128
406	197	0	0	0	0	0	0000203011) 129
406	0	0	1	6007		P	ROP6007 01) 130
124	198	0	1	0	89	79	0000201011) 131
124	250	8	1	0			DRAW 111) 132
402	199	0	0	0	Ο	Ω	0000003017) 133
102		õ	1	7	0	0	SET1 01	131
406 1 00 0	1 2.	U	T	1			01 1 -	, TOA
400,1,80,0	, , , , ;						11	·
406,1,6HPR	(0,0;0,0,0;						31	, 2
406,70,101	,102,103,	104,105,	106,107	,108,109,	,110,111,1	112,11	3,114, 51	° 3
115,116,11	7,118,119	9,120,121	,122,12	3,124,125	5,126,127,	,128 , 1	.29,130, 51	9 4
131,132,13	3,134,135	5,136,137	,138,13	9,140,141	1,142,143,	,144,1	45,146, 51	° 5
147,148,14	9,150,151	,152,153	,154,15	5,156,15 [°]	7,158,159,	,170 , 1	71,172, 51	° 6

173,174,175,176,177,178,179,200,0,1,7;	5P
406,1,6HCHANGE,0,0;	7 P
406,65,0,2,3,4,5,6,7,8,9,10,11,12,13,14,15,16,17,18,19,40,41,42,	9P
43, 44, 50, 51, 52, 53, 54, 102, 103, 104, 105, 106, 107, 108, 109, 110, 111,	9P
112,113,114,115,116,117,118,119,140,141,142,143,144,150,151,152,	9P
153,154,240,241,244,245,246,250,252,253,0,1,11;	9P
406,1,6HWIRALL,0,0;	11P
406,67,0,20,21,22,23,24,25,26,27,28,29,30,31,32,33,34,35,36,37,	13P
38, 39, 45, 46, 47, 48, 49, 55, 56, 57, 58, 59, 120, 121, 122, 123, 124, 125, 126,	13P
127,128,129,130,131,132,133,134,135,136,137,138,139,145,146,147,	13P
148,149,155,156,157,158,159,240,244,246,250,252,253,0,1,15;	13P
406,1,6HSURALL,0,0;	15P
406,28,0,40,41,42,43,44,45,46,47,48,49,140,141,142,143,144,145,	17P
146,147,148,149,240,241,244,246,250,252,253,0,1,19;	17P
406,1,6HUNIQLH,0,0;	19P
406,28,0,50,51,52,53,54,55,56,57,58,59,150,151,152,153,154,155,	21P
156,157,158,159,240,241,244,246,250,252,253,0,1,23;	21P
406,1,6HUNIQRH,0,0;	23P
406, 36, 0, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 46, 47, 56, 57, 120, 121, 122,	25P
123,124,125,126,127,128,129,146,147,156,157,240,241,244,246,250,	25P
252,253,0,1,27;	25P
406, L, SHSUK A, U, U;	27P
400,30,0,30,31,32,33,34,33,30,37,38,39,48,49,38,39,130,131,132,	29P 20D
155,154,155,150,157,150,157,140,149,150,159,240,241,244,240,250,	291
252,255,0,1,51, Ang 1 Sugur B 0 0.	29F 31 D
400, 1, 5000 B, 0, 0, And 32 D 2 3 A 5 6 7 8 9 A1 A2 51 52 102 103 10A 105 106 107	33D DTL
100, 52, 0, 2, 3, 4, 5, 0, 7, 0, 5, 41, 42, 51, 52, 102, 103, 104, 103, 100, 107, 108, 108, 108, 108, 108, 108, 108, 108	330
A06 1 SHWIR A 0 0.	35P
406.36.0.10.11.12.13.14.15.16.17.18.19.43.44.53.54.110.111.112.	37P
113.114.115.116.117.118.119.143.144.153.154.240.241.244.246.250.	37P
252.253.0.1.39:	37P
406.1.5HWIR B.0.0:	39P
406, 25, 0, 5, 15, 25, 35, 42, 44, 47, 49, 52, 54, 57, 59, 105, 115, 125, 135, 142,	41P
144,147,149,152,154,157,159,0,1,43;	41P
406,1,6HREOALL,0,0;	43P
406,20,60,61,62,63,64,65,66,67,68,69,160,161,162,163,164,165,	45P
166,167,168,169,0,1,47;	45P
406,1,6HAUXGEO,0,0;	47P
406,86,0,1,2,3,4,5,6,7,8,9,18,19,20,21,22,23,24,25,26,27,28,29,	49P
41,42,46,47,51,52,56,57,70,71,72,73,74,75,76,77,78,79,100,101,	49P
102,103,104,105,106,107,108,109,120,121,122,123,124,125,126,127,	49P
128, 129, 141, 142, 146, 147, 151, 152, 156, 157, 170, 171, 172, 173, 174, 175,	49P
176,177,178,179,200,240,241,244,246,250,252,253,0,1,51;	49P
406,1,5HMOD A,0,0;	51P
406,88,0,1,10,11,12,13,14,15,16,17,18,19,30,31,32,33,34,35,36,	53P
37,38,39,43,44,48,49,53,54,58,59,70,71,72,73,74,75,76,77,78,79,	53P
100,101,110,111,112,113,114,115,116,117,118,119,130,131,132,133,	53P
134, 135, 136, 137, 138, 139, 143, 144, 148, 149, 153, 154, 158, 159, 170, 171,	53P
172, 173, 174, 175, 176, 177, 178, 179, 200, 240, 241, 244, 246, 250, 252, 253,	53P
0,1,55;	53P
406, I, 5HMOD B, 0, 0;	55P
406,64,0,2,4,5,6,7,8,9,10,11,12,14,15,16,17,18,19,41,42,43,44,	57P
51, 52, 53, 54, 60, 63, 66, 102, 104, 105, 106, 107, 108, 109, 110, 111, 112,	57P
113,114,115,116,117,118,119,141,142,143,144,151,152,153,154,160,	57P 57D
103,100,240,241,244,243,240,230,232,233,0,1,39;	57P 50D
400,1,7mmtrdraw,0,0; 406,66,0,2,4,5,6,7,9,0,10,11,12,14,15,16,17,19,10,41,42,43,44	59P 61D
400,00,0,2,4,3,0,7,0,3,10,11,12,14,13,10,17,10,13,41,42,43,44, 51 52 53 54 60 63 66 100 102 104 105 106 107 108 108 110 111	61 D
$112 \ 113 \ 114 \ 115 \ 116 \ 117 \ 118 \ 119 \ 141 \ 142 \ 143 \ 143 \ 144 \ 151 \ 152 \ 153 \ 154$	61 P
160 163 166 200 240 241 244 245 246 250 252 253 0 1 63	61 P
406.1.15HWIRDRAW+REF PT .0.0:	63P
406.67.0.2.3.4.5.6.7.8.9.10.11.12.13.14.15.16.17.18.19.40.41.42.	65P
43, 44, 50, 51, 52, 53, 54, 100, 102, 103, 104, 105, 106, 107, 108, 109, 110.	65P
111.112.113.114.115.116.117.118.119.140.141.142.143.144.150.151.	65P
152,153,154,200,240,241,244,245,246,250,252,253,0,1,67;	65P
406,1,14HWIRALL+REF.PT.,0,0;	67P
406,17,0,201,240,241,242,243,244,245,246,247,248,249,250,251.	69P
252,253,254,0,1,71;	69P
406,1,9HSEC.START,0,0;	71P
406,199,0,1,2,3,4,5,6,7,8,9,10,11,12,13,14,15,16,17,18,19,20,21,	73P
22,23,24,25,26,27,28,29,30,31,32,33,34,35,36,37,38,39,40,41,42,	73P
43,44,45,46,47,48,49,50,51,52,53,54,55,56,57,58,59,60,63,66,70,	73P
71,72,73,74,75,76,77,78,79,81,82,90,91,92,93,100,101,102,103,	73P
104,105,106,107,108,109,110,111,112,113,114,115,116,117,118,119,	73P
120,121,122,123,124,125,126,127,128,129,130,131,132,133,134,135,	73P
136,137,138,139,140,141,142,143,144,145,146,147,148,149,150,151,	73P

152 153 154 155 156 157 158 159 160 163 166 170 171 172 173 174	73P	84
132,133,134,133,130,137,130,133,100,103,100,171,172,173,174,	751	04
175,176,177,178,179,200,201,202,203,204,205,206,207,208,209,210,	73P	85
211,212,213,214,215,216,217,218,219,220,221,222,223,224,225,226,	73P	86
227 228 229 230 231 232 233 234 235 236 237 238 239 240 241 244	730	87
227,220,227,230,231,232,233,233,233,230,237,230,237,230,237,240,241,244,	751	07
246,250,252,253,0,1,75;	/3P	88
406,1,6HPARAMS,0,0;	75P	89
124,1.0,0.0,0.0,0.0,0.0,1.0,0.0,0.0,0.0,0.0	77P	90
	70D	Q 1
124,1.0,0.0,0.0,0.0,0.0,1.0,0.0,0.0,0.0,0.0	7 9 E	51
402,1,1,79,131,0,0;	81P	92
406,1,1,0,0;	83P	93
124.1.0.0.0.0.0.0.0.0.0.1.0.0.0.0.0.0.0.0.	85P	94
	070	05
406,1,0,0,0;	0/2	95
410,1,1.0,0,0,0,0,0,0,1,81,2,83,87;	89P	96
404,1,89,1.0E+05,1.0E+05,0,0,1,93;	91P	97
406 2 2 0 <u><u><u></u></u><u></u><u></u><u></u><u></u>0<u></u><u></u><u></u><u></u><u></u><u></u><u>0</u><u></u><u></u><u></u><u>0</u><u></u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u><u>0</u><u></u></u> <u>0</u> <u></u> <u>0</u> <u>0</u>	03D	98
	221	50
314,100.0,53.33336114883423,53.33336114883423,/HCOLORI2,0,0;	95P	99
314,86.666667461395264,0.0,66.666668653488159,7HCOLOR14,0,0;	97P	100
314,93.333333730697632,0.0,0.0,7hCoLOR15,0,0;	99P	101
	1010	102
	1011	102
1.0, 1.0, 1.0, 1.0, 1.0, 1.0, 1.0, 1.0,	101P	T03
1.0,1.0,2620.660337813827,-741.77987839884747,	101P	104
1753.7369910376553.2639.4709781390311746.64767231685596.	101P	105
	1010	106
1/41.52015/5509104,2012.75/192/052055,7745.5594020510297,	1011	100
1745.1203595594527,2632.0672165209398,-748.12478890822501,	TOID	T0./
1733.6549005850775,2605.2768208125667,-745.28121211978828,	101P	108
1736 1037283247665 2625 0332407982714 -749 5414510328194	101P	109
	1011	110
1/25.42253/1883663,2598.2840/82645162,-/46.928//53/61/01/,	TOIP	110
1726.6767404740067,2618.4289414884206,-750.88429546278371,	101P	111
1716.8177357803215.2591.8527293460966748.4827516953136.	101P	112
1716 0427603036404 2612 3250225112700 _752 13754010070034	101D	112
1/10.042/003930404,2012.3230223112/00, -/32.13/340100/0034,	TOTE	115
1707.837702667734,2586.0622796869411,-749.92070228417379,	101P	114
1706.6316269498975,2606.8028308588177,-753.28310858515192,	101P	115
1698 5011333688924 0 0 1 0 0 0 1 0 0 1 103.	101P	116
	1011	117
406,2,1,1,0,0;	103P	11/
126,10,5,1,0,1,0,0.0,0.0,0.0,0.0,0.0,0.0,1.0,1.0	105P	118
2.0.2.0.2.0.2.0.2.0.2.0.1.0.1.0.1.0.1.0.	105P	119
1 0 0 23245663602028367 0 65540340508675776 0 0	105D	120
1.0,0.22245005025220507,0.0534534050075770,0.00	1051	101
0.24494091570472551,0.5794784437124092,0.0,0.25628879766477253,	105P	121
0.50343448244869449,0.0,0.26644767965591071,0.42736374550523398,	105P	122
0.0.0.27539924956721623.0.35127002095234117.0.0.	105P	123
0.29316405422363190.0.27515750690530054.0.0.0.29415030967277202	105D	124
0.26316403422303169, 0.27313730669339034, 0.0, 0.26413039607277292,	IUJE	124
0.26548841998961314,0.0,0.28511767309704522,0.25581904492108354,	105P	125
0.0,0.28606587749644885,0.24614938168980172,0.0,	105P	126
0 28699501187098375 0 23647943029576768 0 0 0 28790507622064992	105P	127
	1051	100
0.22680919073898143,0.0,0.0,2.0,0.0,0.0,1.0,0,0;	105P	128
126,5,5,1,0,1,0,0.0,0.0,0.0,0.0,0.0,0.0,1.0,1.0	107P	129
1.0,1.0,1.0,1.0,1.0,1.0,-1.1102230246251565E-16,	107P	130
0 17905783724057648 0 0 20208529840797829 0 21184062860381359	107Þ	1 3 1
0.1/90/05/2409/040,0.0,0.2222040/9/023,0.21104002000501555,	1071	100
0.0,0.40324203327511476,0.24573031692104391,0.0,	10/P	132
0.60334072730673416,0.28078840664032911,0.0,0.80230968779290301,	107P	133
0.31708079404494116,0.0,1.0000000000000007,0.35466240391881665,	107P	134
0 0 0 28571233372997135 0 98240689650936364 0 0 0 0 1 0 0 0	1070	135
	1000	100
126,5,5,1,0,1,0,0.0,0.0,0.0,0.0,0.0,0.0,1.0,1.0	1095	136
1.0,1.0,1.0,1.0,1.0,1.0,0.98192028465554726,0.22237481410268387,	109P	137
0.0,0.98269327479807422,0.37789942943733612,0.0,	109P	138
0 98366419421193763 0 53342386576180689 0 0 0 98454857636867932	109P	139
	1000	140
0.88894836010605798, 0.0, 0.98558424076097029, 0.8444727295665975,	1095	140
0.0,0.98636055026224656,0.99999734287303721,0.0,	109P	141
0.16587167932528243,0.76847136763512491,0.0,0.0,1.0,0,0;	109P	142
126 5 5 1 0 1 0 0 0 0 0 0 0 0 0 0 0 0 1 0 1	111D	1/13
	1111	145
1.0,1.0,1.0,1.0,1.0,1.0,0.985363998812/03/6,0.8199485866/838322,	TTTL	144
0.0,0.83707180288432625,0.78549124371198509,0.0,	111P	145
0.68755906050604521,0.75183487849241715,0.0.0.5369075393959869.	111P	146
0 7189650513986523.0 0.0 38518183026886199 0 68686180561249666	 111 D	1/7
	+ + + + + + + + + + + + + + + + + + +	1 4 0
0.0,0.232434309133223,0.65550/10/32156214,0.0,0.0,1.0,0.0,0.0,	TTTL	148
1.0,0,0;	111P	149
102,4,105,107,109,111,0,0;	113P	150
	1150	151
	11JP	TOT
2.0,2.0,2.0,2.0,2.0,2.0,1.0,1.0,1.0,1.0,1.0,1.0,1.0,1.0,1.0,1	115P	152
1.0,2601.4625231380633,-748.21563064534712,1721.5291120921393,	115P	153
2604.1894023485283747.69105077977986.1724.9995472943076	115P	154
	1155	101
2000.9/33940040300,-/4/.131102649430931,1/28.42223330/3292,	TTDP	TDD
2609.8109586822575,-746.59604688395154,1731.7989722811562,	115P	156
2612.6976801659707,-746.02588703240133,1735.1309352249853,	115P	157
2615.6305472708168745.44110273641093.1738.4188074122919	115P	158
2616 001708/2200/ _7/5 26700500710261 1720 02/000620171	1150	1 5 0
2010.001/004220904,-743.30/09300/18301,1/38.83488890281/1,	TTOP	T28
2616 3736017456131.=745 29285793464953.1739 2502757312011.	115P	160

2616.7462391191771,-745.21838417275285,1739.6649542837099,	115P	161
2617.1196101996197,-745.14367544209688,1740.078928710612,	115P	162
2617.4936955653984,-745.06873699751236,1740.4922152774345,0.0,	115P	163
2.0,0.0,0.0,0.0,0;	115P	164
126,5,5,1,0,1,0,0.0,0.0,0.0,0.0,0.0,0.0,1.0,1.0	117P	165
1.0,1.0,1.0,1.0,1.0,1.0,2613.7236749491458,-743.35371835426668,	117P	166
1745.8934003500985,2616.3408065564863,-744.55413329069256,	117P	167
1742.143863388405,2618.960298601432,-745.73895663563883,	117P	168
1738.3909515536966,2621.5822868514351,-746.90681048015676,	117P	169
1734.6344594063166,2624.2069428993518,-748.05609461794268,	117P	170
1730.874140020285,2626.8345078809853,-749.18486625125206,	117P	171
1727.1096605078169,0.2879484711334489,0.98261164072408724,	117P	172
-0.82000696081867042,-5.8451089264416334E-06,	117P	173
-0.57235354823287532,0,0;	117P	174
126,5,5,0,0,1,0,0.0,0.0,0.0,0.0,0.0,0.0,1.0,1.0	119P	175
1.0,1.0,1.0,1.0,1.0,1.0,2631.0788601203235,-748.17673542490513,	119P	176
1732.8006056022314,2625.6015116315516,-749.28788849145542,	119P	177
1726.4089751149004,2620.3758703547282,-750.35771471811756,	119P	178
1719.7922558463156,2615.429929355194,-751.37749019737487,	119P	179
1712.9526133416707,2610.801757236386,-752.3405417025167,	119P	180
1705.8894656800142,2606.5200145680296,-753.23723194092361,	119P	181
1698.6121530281373,0.16589786618932054,0.76849627089981454,	119P	182
-0.82000696081867042,-5.8451089264416334E-06,	119P	183
-0.57235354823287532,0,0;	119P	184
126,5,5,1,0,1,0,0.0,0.0,0.0,0.0,0.0,0.0,1.0,1.0	121P	185
1.0,1.0,1.0,1.0,1.0,1.0,2611.6583758345132,-752.16484110141641,	121P	186
1706.9202960744065,2609.6152747931365,-751.40561626746,	121P	187
1709.847421042806,2607.5741834720625,-750.63000354408484,	121P	188
1712.7716743387873,2605.5349911961612,-749.83916048256788,	121P	189
1715.6931958309951,2603.4976303146464,-749.03409956152166,	121P	190
1718.61210377916,2601.4620295148675,-748.2157254693733,	121P	191
1721.5284844870639,0.0,1.0,-0.82000539781986614,	121P	192
5.6468497980355359E-06,-0.57235578752590266,0,0;	121P	193
102,4,115,117,119,121,0,0;	123P	194
142,0,101,113,123,0,0,0;	125P	195
144,101,1,0,125,0,1,129;	127P	196
406,2,2,2,0,0;	129P	197
124,1.0,0.0,0.0,0.0,0.0,1.0,0.0,0.0,0.0,0.0	131P	198
402,1,127,0,0;	133P	199
S 3G 4D 134P 199	Т	1

Nas-file

Structure of a Nastran nas-file generated by ADRIAN

1 2 3 4 5		3300.874 3301.214 3297.169 3293.124 3289.081	-574.325 -576.405 -576.740 -577.080	52116.201 52114.595 52114.145 52113.687 52113.223	L 5 7 3	100 100 100 100 100
1 2 3 4 5	1 1 1 1	24 27 22 29 20	25 28 23 30 21	26 23 28 21 30	27 24 29 22 31	
· PSHELL E 1 MAT1 Ma	Property 1 aterial	0.80	1	1.000000)1	
1	210000.0)	.3000000	.78500-8	3	
4000 4001 4002 4003 4004		3185.794 3185.464 3184.719 3188.229 3191.689	-611.249 -608.102 -605.569 -605.506	02040.905 22040.789 02042.535 52042.242 22041.815		100 100 100 100 100
3903 3904 3905 3906 3907	2 2 2 2 2	4003 4001 4009 4008 4005	4010 4002 4010 4009 4006	4001 4003 4004 4004 4008		
PSHELL 2	Property 2	1.5	1	1.000000)1	
MAT1 Ma 2	aterial 210000.0)	.3000000).78500-8	3	
7745 7746 7747 7748 7749	3	3311.263- 3310.681 3310.099 3309.517 3308.935	612.6272 -610.066 -607.506 -604.946	2048.716 52052.196 52055.677 52059.157 52062.636	5 7 7 5	100 100 100 100 100
7709 7710 7711 7712 7713	3 3 3 3 3	7917 7917 7918 7837 7919	7835 7836 7834 7838 7833	7836 7837 7835 7920 7834	7917 7917 7918	
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\$Default PSHELL Property

PSHELL 3	3	3	1.2	1	1.000000	1	
+0007867							
+0007867							
\$Default	MAT1 Ma	iterial					
MAT1 3	3	210000.0		.3000000	.78500-8		
+0007870							
+0007870							
PSOLID 1	.0000000	1					
CHEXA	9000000	1000000	8000001	8000002	8000005 8	3000007 8	3000000
800003+							
+	8000004	8000006					
GRID	8000001		2990.33	-666.63	1959.08		100
GRID	8000000)	2990.28	-668.65	1960.26		100
GRID	8000002		2986.83	-667.53	1957.40		100
GRID	8000003	3	2986.78	-669.54	1958.57		100
GRID	8000005)	2984.92	-665.74	1960.41		100
GRID	8000004	L	2984.88	-667.75	1961.58		100
GRID	8000007	,	2988.42	-664.84	1962.09		100
GRID	8000006	-	2988.38	-666.85	1963.27		100
CHEXA	9000001	10000000	8000009	8000012	8000016	8000019	8000010
8000013+							
+	8000015	8000017					
CHEXA	9000002	10000000	8000010	8000013	8000015	8000017	8000008
8000011+							
+	8000014	8000018					
GRID	8000009)	3009.81	-658.06	1971.69		100
GRID	8000010)	3009.78	-659.22	1972.38		100
GRID	8000008	2	3009.74	-661.22	1973.58		100
GRID	8000012)	3013.33	-657.16	1973.31		100
GRID	8000013	-	3013 31	-658 32	1974 01		100
GRID	8000011		3013 27	-660 32	1975 21		100
GRID	8000016	-	3015 19	-659 01	1970 30		100
GRID	8000015		3015 17	-660 17	1971 00		100
GRID	8000014		3015 12	-662 17	1972 20		100
GRID	8000014)	3011 66	-659 90	1968 67		100
GRID	8000017	,	3011 64	-661 06	1969 37		100
GRID	8000018	2	3011 59	-663 06	1970 57		100
CHEXA	9000010	,	8000021	8000024	8000027	8000029	8000022
8000025+	5000000	100000000	0000021	0000021	0000027	0000029	0000022
+	8000028	8000031					
CHEXA	9000020	10000000	8000022	8000025	8000028	8000031	8000020
8000023+	2000001	10000000	0000022	0000025	0000020	00000001	0000020
+	8000026	8000030					
CRID	8000020	00000000	3037 25	-619 97	1985 89		100
GRID	8000021	-)	3037.23	-651 11	1986 60		100
CRID	8000022		3037.23	-653 10	1087 83		100
CRID	0000020		2040 02	-610 07	1007.03		100
CRID	0000024		3040.82	-650 22	1000 15		100
CRID	0000023)	2040.75	-652 20	1000.13		100
GRID	0000023		2040.75	-652.20	1001 12		100
GRID	0000027	•	2042.01	-050.97	1005 11		100
GRID	0000020	-	3042.39 2042 EE	-652.12	1905.14		100
GRID	8000026		3042.55	-654.10	1986.37		100
GRID	0000023	,	2020 00	-001.00 -652.01	1002 E0		100
GKID	0000031		2020 00	-000.UI	1001 01		100
GKID	0000030	10000000	3030.99	-000007	1904.01	0000040	T00
CHEXA	9000005	100000000	8000034	800003/	8000040	8000043	8000033
8000036+ '	0000000	0000040					
+	8000039	8000042					
·····	000001		0000001	1	0001150	100	
LDES	0000001		0000001	123	.0031132	123	
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+		123	5437	7.8881174	l	123	3	5440	.0405	5646	5	123	
5439 RBE3	8000	000		8000000)	123	8.006	0098	3	123	3		
341.0087 +	406+	123	342	2.5838278	3	123	3	345	.4014	1219)	123	5
344 RBE3	8000	002		8000002	2	123	8.0028	8388	3	123	3		
5433.053 +	36406+	123	5434	1.8960970)	123	3	5437	.0474	1236	5	123	1
5436 RBE3	8000	003	5	8000003	3	123	3.0074	4442	2	123	3		
338.0103 +	8219+	123	339	9.5706661	-	123	3	342	2.4115	5678	3	123	
341 RBE3	8000	005		8000005	5	123	8.923	5667	,	123	3		
5438.024 + 5435	19334+	123	5437	7.0013538	}	123	3	5434	.0501	L461	-	123	5
CORD2R 2098.55	1		0	3304.23	-592.	60	2098	.66	3303.	.24	-592.	68	
CONM2	180.5 50000	3 01	592.52 6000001	-2098.77 1	7 0.005	5							
CORD2R 2090.33	19.7 2		0	19.7 3081.36	-454.	81	2090	.39	39.0 3081.	.59	-455.	78	
CONM2	1360. 50000	55 02	453.84 6000002	-2090.46 2	5 0.005	5			30 0				
CORD2R	3		0	2958.77	-663.	84	1985	.36	2959.	.65	-663.	62	
CONM2	801.5	1 03	664.05 6000003	-1984.93	3 0.005	5			2.0.0				
GRID RBE2	19.7 60000 70000	01 01	6000001	19.7 3304.23 123456	-592. 8768	598	32098 8767	.66	39.0 8766		8765		8764
+	8763		8712	8595	8594		8593		8541		8406		8405
+	8404		8403	8402	8401		8338		8337		8336		8212
+	8211		8210	8209	8208		8207		8158		8157		7749
+	7748		7747	7746	7745		1819		1818		1817		1816
+	1815		1814	1813	1812		1719		1718		1717		1716
+	1563		1363	1362	1361		1178		1173		1172		899
+	898		897	892	621		610		609		608		607
+	602		601	152	151		150		149		148		134
+	133		132	131	130		129		128		127		126
+ GRID RBE2	125 60000 70000	02 02	99 6000002	2 3081.36 123456	1 -454. 10009	809)2090 10008	.39 8	10007	7	10006	5	10005
+	10001		9997	9584	9583		9582		9581		9580		9579

+	9578	9577	9576	9575	9574	9573	9572	9571
+	9570	9569	9568	9567	9537	9499	9228	9227
+	9226	9225	9224	9223	9222	9221	9220	9116
+	9115	6007	6006	6005	6004	6003	6002	5949
+	5948	5871	5870	5869	5847	5824	5823	5822
+	5810	5809	5808	5807	5806	5805	5804	5803
+	5800	5799	5784	5783	5782	5781	5766	5765
+	5764	5672	5671	5571	5570	5569	5568	5549
+ GRID RBE2	5548 6000003 7000003	5547 6000003	5543 2958.77 123456	5542 -663.835 11344	51985.36 11343	11342	11341	11340
+	11339	11338	11195	11194	11193	10685	10684	10683
+	10682	10681	10680	5426	5425	5424	5423	5308
+	5307	5249	5244	5243	5242	5236	5235	5227
+	5226	5225	5093	5092	5091	5090	5089	5088
+	4993	4992	4991	4900	4899	4898	4897	4896
+	3155	3154	3153	3152	3151	3150	3149	3025
+	3024	3023	3022	2985	2984	2918	2885	2795
+	2530	2529	2528	2352	2351	1566	1565	1564
+	1269	1268	904	903	902	890	712	710
+	245	244	243	242	241			

Development of Joint Analysis Software - Appendix C

Appendix D - Results

HTML result documents and pictures of freqency animations of the first mode for five joints in Volvo S80-series:

Result from ADRIAN job:

Fri Mar 15 16:16:37 JobID: 16141 Model: P23_A_UPPER





Fri Mar 15 16:23:49 JobID: 16200 Model: P23_A_MIDDLE



Result from ADRIAN job:

Fri Mar 15 16:31:21 JobID: 16293 Model: P23_B_UPPER



Result from ADRIAN job:

Fri Mar 15 16:38:10 JobID: 16340 Model: P23_B_LOWER



Result from ADRIAN job:

Fri Mar 15 16:46:28 JobID: 16434 Model: P23_C_UPPER













P23_B_LOWER



Appendix E - ADRIAN Help section

ADRIAN help 1.1

Catia

• Joint structure

IGES conversion

• <u>IGES V1.1</u>

ADRIAN

- <u>File menu</u>
- <u>Complete menu</u>
- <u>Pre menu</u>
- <u>Nastran menu</u>
- <u>Post menu</u>
 - Import pch (View stiffness) menu
 - Import op2 (freq. animation) menu
- <u>Options menu</u>
 - <u>Settings menu</u>
 - <u>Advanced</u>

This document was generated on 1 March 2002

Joint structure

This section describes what to export from Catia. To use ADRIAN successfully it is important to use the syntax described below. It is also very important that the model exported is clean and therefore only consist of the components described below:

- Each sheet shall be placed in a separate Catia set
- A separate set consisting of **planes** describing the beam ends must be existent
- Sets for the welds

Sheets

Syntax

If the model for example consists of three sheets put each sheet in a separate set. Use the following syntax when creating a setname: **NAME:TSHEETTHICKNESS**, a setname may consist of maximum 30 characters. **NAME** is a unique name for each set, for example SET1, SET2 and SET3. The second part, **SHEETTHICKNESS**, is the sheet thickness given in millimetres in the range 0.5-2.5 mm. The sheet thickness is optional, if no value is given ADRIAN uses a default sheet thickness of 1.0 mm. The following setnames are examples of valid ADRIAN setnames: SET1:T2.1, SET2:T0.9 and SET3.

Restrictions on geometry

To obtain the best prestanda for Adrian the model should have faces on every surface. Make sure faces are applied on every surface in Catia and simply erase all surfaces. This will most often erase all surfaces which have not a corresponding face. Sometimes surfaces will still be existent in the model which are not necessary for describing the model. To avoid this it might be a good thing putting on a skin and copying the surfaces and faces of interest to an empty Catia model. If a model is imported into Adrian which consist of surfaces which has not a corresponding face, Adrian gives a user warning. If the model has faces on each surface, click Remove. This will remove all surfaces which do not belong to the model. If the model consist of both surfaces and faces, click Accept. Note that the Accept interaction can not garantee as good mesh quality as a model with face on each surface and that there may be remaining surfaces that do not belong to the model. When using a model with surfaces which have not a corresponding face, it is thus important that there are only surfaces necessary for descibing the model present.

Planes

A separate set consisting of planes describing the joint ends must be existent. ADRIAN looks for a set consisting of three or four planes, depending on joint type. The setname must be unique and consist of maximum 30 characters. There must be nothing else than these planes in the plane set.

Welds

In order to weld two sets together, simply create a new set named **NAME:NAME:DDIAMETER**, where **NAME** is the name part of the sheet set and **DIAMETER** is the weld diameter given in millimetres and in the range 4.0-9.9 mm, for example: SET1:SET2:D4.5. Note that it is also possible to weld three or four sets together, for example: SET1:SET2:SET3:D6.3 or SET1:SET2:SET3:SET4:D5.5. The weld diameter is optional; it is possible to create a weld set with the syntax **NAME:NAME**; for example SET1:SET2. If this is done, ADRIAN uses a default weld

diameter of 4.5 mm. As the set is created, just put points marking the position of the welds. If, for some reason, Adrian fails to create a weld, the user get a warning message telling which point not resulting in a correct weld. Make sure weldpoint is in a logic position and that it belongs to a correct set, named after the sheets being welded together Note that there must be nothing else than points in a weld set. Note also that there is no limit in how many weld sets that can be existent in a model.

IGES V1.1

In order to convert the Catia model to IGES, type >iges

at the command prompt.

- Choose conversion <1> Catia to IGES
- Choose the path to the file that is to be converted
- Pick your Catia model file
- Choose what directory to place the resulting IGES-file in
- Enter name of the IGES file (Just pressing enter gives the Catia model name)
- Choose what to transfer <1> SPACE (exter process GEOM)
- Choose layer option <1> All layers to one iges file

Go to the <u>table of contents</u>.

File menu

Left click on the File menu button. A scrolldown menu appears with an Exit-button. If Exit is chosen, a pop-up menu appears asking the user to confirm the choice. To exit ADRIAN press OK.

Go to the <u>table of contents</u>.

Complete menu

Complete processing does all components of ADRIAN in one sequence; Pre-processing, Nastran calculation and Post-processing. The igs-file must have the structure described in the <u>Joint</u><u>structure</u>-section. Left click on the Complete menu button. A scrolldown menu appears with an Import igs-button. If Import igs is chosen, a file manager appears. Here it is possible to navigate through the file system. Chose an appropriate IGES-file and press ok. This will start the processing. Note that Complete processing uses pre-processing settings and Nastran settings defined in the <u>Settings menu</u>. For Complete processing it is possible to set what kind of post-processing ADRIAN should do.

While running information on the job is displayed in the Output window. This information consists of a jobid, the contents of the igs-file, etc. As pre-processing is finished "Pre-processing done" is printed to the output window. The nastran-file can now be found in the \$HOME/ADRIAN_ROOT/ JOBNAME directory, where JOBNAME is the name of the igs-file and a jobid consisting of 5 digits separeted by the character '_', for example myjoint_12352. Together with the Nastran-file is also a .post file used for displaying geometry in the <u>Stiffness postprocessor</u>. Now the nastran-file is submitted to a server. As Nastran has finished all its output files are copied to

\$HOME/ADRIAN_ROOT/**JOBNAME**. The three most important output files are f06-file, which contains error information about the calculation, the op2-file and pch-file, which are input files for the 2 different post-processing routines. As <u>Nastran</u> processing is finished the f06-file is scanned for errors. If a fatal error occurred during the calculation ADRIAN prints an error message to the output window and then returns to the main window. If a warning or a error occurred ADRIAN reports the number of warnings and errors to the output window and then prints "Nastran done" to the same window. If no fatal errors occurred the complete process continues with the post-processing. The available post processors are <u>Stiffness</u> and <u>Frequency</u> visualization

Pre menu

From the Pre menu, pre-processing is carried out. Pre-processing in ADRIAN converts a joint in the igs-format to the Nastran format. The igs-file must have the structure described in the <u>Joint</u> <u>structure</u> -section. Left click on the Pre menu button. A scrolldown menu appears with an Import igs-button. If Import igs is chosen, a file manager appears. Here it is possible to navigate through the file system. Choose an appropriate IGES-file and press ok. This will start the processing. Note that Pre-processing uses the pre-processing settings defined in the <u>Settings menu</u>. Information about the job is displayed in the Output window. This information consists of a jobid, the contents of the igs-file, etc. As post-processing is finished "Post-processing done" is printed to the output window. The nastran-file can now be found in the <u>\$HOME/ADRIAN_ROOT/JOBNAME</u> directory, where JOBNAME is the name of the igs-file and a jobid consisting of 5 digits separated by the character '_', for example myjoint_12352. The .post file, which is also found in this directory, is used in the <u>stiffness postprocessing</u>.

<u>Nastran menu</u>

From the Nastran menu, Nastran jobs are submitted out. Left click on the Nastran menu button. A scrolldown menu appears with an Import nas-button. If Import nas is chosen, a file manager appears. Here it is possible to navigate through the file system. Choose an appropriate Nastran-file and press ok. This will start the processing. In order to use Nastran calculation in the Adrian program, the Nastran-file must be produced by the Adrian preprocessor. Note that Nastran uses the Nastran settings defined in the <u>Settings menu</u>.

Information on the job is displayed in the Output window. Note that the jobid displayed is a local jobid used by Nastran. The nastran-file is submitted to a server all Nastran output files are copied to the same directory as the original nastran-file. The three most important output files are the f06-file, which contains error information about the calculation, and the op2-file and pch-file, which are input files for the 2 different post-processing routines. As Nastran processing is finished the f06-file is scanned for errors. If a fatal error occurred during the calculation ADRIAN prints an error message to the output window and then returns to the main window. If a warning or a error occurred ADRIAN reports the number of warnings and errors to the output window and then prints "Nastran done" to the same window. If no fatal errors occurred it is possible to start the post-process.

Time-out

ADRIAN waits for Nastran to finish. If Nastran has not terminated after 2 hours, ADRIAN stops waiting. This may be due to a long queue on server. Note that if Nastran terminates after these two hours the user will still get Nastran output files copied to the directory of the submitted job. The user will then have to invoke ADRIAN postprocessing manually if running <u>Complete processing</u>. Note that ADRIAN does not give the user information if Nastran for any reason fail to return the f06-file. If this is the case, ADRIAN will wait for 2 hours and exit.

Displaying the queue

In order to display the queue for the submitted Nastran job type

>. /vcc/apps/lsf/mnt/conf/profile.lsf

>xlsbatch

at your unix prompt. To locate your Nastran job sort the queue by jobid. The Nastran jobid for the submitted job is found in ADRIAN output window (Nastran Job <xxxxx> is submitted to queue <csm quick>.)

Post menu

In the Options menu it is possible to enter the <u>import pch</u>-menu and the <u>import op2</u>-menu

Post pch-menu

From the Post pch menu, stiffness visualisation is carried out. Left click on the Post menu button. A scrolldown menu appears with an Import pch (view stiffness)-button. If Import pch is chosen, a file manager appears. Here it is possible to navigate through the file system. Choose an appropriate pch-file and press ok. This will start the processing. The pch-postprocessing also uses the .post file which must be existent in the same directory as the .pch file.

Information on the processing is displayed in the Output window. As post-processing is finished "Postprocessing pch done" is printed to the output window. Netscape is opened and the results are shown. The result consists of a picture of the reference view and pictures of each joint end. The sheet thicknesses and weld diameters used are given. In the joint end-pictures, the green axis represent the minimum stiffness direction and the red axis represent the maximum stiffness direction. Beneath each picture , three different stiffnesses are displayed. The maximum stiffness is below the red arrow, the minimum is below the green arrow. The third stiffness represent rotation in the "picture plane". Information on what model processed and when it was sent to Nastran calculation is also displayed.

Post op2-menu

From the Post op2 menu, frequency animation is carried out. Left click on the Post menu button. A scrolldown menu appears with an Import op2 (freq. animation)-button. If Import op2 is chosen, a file manager appears. Here it is possible to navigate through the file system. Choose an appropriate op2-file and press ok. This will start the processing. Information about the processing is displayed in the Output window. As op2 processing is finished "Post-processing op2 done" is printed to the output window. Animator is now started.

Animator

Changing the view

In order to reposition the view use: **Ctrl+leftmousebutton**-Rotate the picture with the mouse **Ctrl+middlemousebutton**-Translate the picture with the mouse

Ctrl+leftmousebutton+middlemousebutton-Zoom in by moving mouse downwards, Zoom out by moving mouse upwards. Ctrl+rightmousebutton-Rotate the picture around an axis in he viewplane with the mouse Ctrl+F1-top view Ctrl+Shift+F1-bottom view Ctrl+F2-right view Ctrl+Shift+F2-left view Ctrl+F3-front view Ctrl+Shift+F3-rear view

Viewing node displacement

In order to view the node displacement in a color map choose Utils->sty->int->dno->all.

Changing frequency mode

It is possible to toggle through the first ten modes by using the F1-F10 buttons. To obtain the next ten modes press F12 and to obtain the previous ten modes press F11. There are totally 13 modes available for viewing.

Changing animation speed

Change the animation speed by entering the Utils->ani->spe menu.

Changing animation displacement

Change the animation displacement by entering the Utils->dis->sca menu.

Disable animation

In order to disable animation press **Ctrl+Delete**. For full documentation please refer to the Animator Help guide.

Go to the <u>table of contents</u>.

Options menu

In the Options menu it is possible to enter the <u>Settings menu</u> and the <u>Advanced</u> settings menu.

<u>Settings menu</u>

The Settings menu is divided into three parts Pre-processing settings, Nastran settings and Complete Process settings.

Pre-processing settings

Mesh size

Point on the scale and left-click in order to change the mesh size used by Ansa. Possible values for the mesh size are between 1 and 20 mm.

Use Catia sheet thickness

If the box is checked, ADRIAN looks for the sheet thickness defined in the setnames in Catia. The thickness should be given according to the syntax **NAME**:**TSHEETTHICKNESS**; a setname may consist of maximum 30 characters. **NAME** is a unique name for each set and **THICKNESS** is the sheet thickness of the sheet given in millimetres and in the range 0.5-2.5 mm, for example: SET1:T1.5. If the box is unchecked, ADRIAN lets the user define the sheet thickness for each sheet set. In the first part of the pre-processing ADRIAN identifies the contents of the igs-file. A pop-up menu occurs with all the existing geometry sets. Enter the values desired and press the OK-button.

Use Catia weld diameter

If the box is checked, ADRIAN looks for weld diameters defined in the setnames in Catia. The diameter should be given according to the syntax **NAME:NAME:DDIAMETER**, where **NAME** is the name part of the sheet set and **DIAMETER** is the weld diameter given in millimetres and in the range 4.0-9.9 mm, for example: SET1:SET2:D4.5. Note that it is also possible to weld three or four sets together, for example: SET1:SET2:SET3:D6.3 or SET1:SET2:SET3:SET4:D5.5. If the box is unchecked, ADRIAN lets the user define the weld diameter. In the first part of the pre-processing ADRIAN identifies the contents of the igs-file. A pop-up menu occurs with all the existing weld sets. Enter the values desired and press the OK-button.

Nastran settings

Max ratio

Nastran computes the ratios of the terms on the diagonal of the stiffness matrix to the corresponding terms on the diagonal of the triangular factor. If, for any row, this ratio is greater than maxratio, the matrix will be considered to be nearly singular. The default value for maxratio is 1E7.

Bailout

If the box is checked, the job is submitted to Nastran with the parameter BAILOUT set to -1. This makes Nastran continue processing if near singularities are found in the stiffness matrix. If a an Adrian job result in errors when calculating in Nastran, checking this box might be a good idea. This will make Adrian produce a result. Note that the solution may be of poor quality.

Queue

Determines what queue the nastran job is submitted to. For normal jobs the default queue csm_quick should be sufficient. This sets the maximum processing time to 10 min. For very large joint geometries, choose the csm_small option.

Complete process settings

Do stiffness post-processing

If the box is checked, ADRIAN automatically launches the stiffness matrix post-processing as a complete process job is submitted.

Do frequency post-processing

If the box is checked, ADRIAN automatically launches the frequency animation post-processing as a complete process job is submitted.

Advanced

From the Advanced settings menu it is possible to set ADRIAN in Ansa user interactivity mode. Left click on the Options menu button. A scrolldown menu appears with an Advanced-button. If Advanced is chosen, a pop-up menu appears. If the User interactivity mode is set Ansa will process the different geometry sets one at a time. This gives the possibility to manipulate the geometry in Ansa, with functions like **Paste**, **Delete Double** etc. It is very important that the user exports a mesh to a nas-file. Note that the filename of the nas-file must be the same as the igs-file for example: **INPUT:Set3.**igs, **OUTPUT:Set3.**nas.

Use the flow of interactions described below: As the igs-model have been read into Ansa, use the **TOPO** menu deck to manipulate the surfaces. Use the **MESH** menu deck to create a mesh on the surfaces, and to manipulate the mesh created by Ansa. After having created a suitable mesh. choose **FILE->OUTPUT->NASTRAN** Type the name of the nastran-file in the Selection-box. The name **MUST** be the same as the file read in, with the suffix changed from .igs to .nas. The name of the file read in can be found in the Ansa Text Dialog. If for example SET1.igs is read in, type SET1.nas in the Selection box. Do **NOT** change the (temp) directory Ansa suggests!. Click the OK button. Press OK in the Output Parameter dialog Press Cancel in the Nastran Dech Parameters dialog Choose **FILE->QUIT** Repeate the action above for each set read in.

Appendix F – Structural diagrams

Pre-processing



Nastran calculation



Post-processing


Complete process

